

## MINOR CHANGE APPROVAL

### 10080105

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### 4DRIVE & AVIATION MACIEJ SZANTER

SLONECZNA 54M  
05-500 STARA IWICZNA  
POLAND

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** EASA.IM.R.121

EASA.IM.R.507

**Type Certificate Holder:** ROBINSON HELICOPTER COMPANY

**Type:** R44 and R66

**Model:** R44, R44 II, R66

#### Description of Design Change:

Minor design change to change seat covers, upholstery covers and add new pockets on door panels with certified leather, and alcantara covers using materials/practices described in Robinson IPC/AMM and AC43.13

#### EASA Certification Basis:

The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 08 September 2022



Fabrice LEGAY  
Section Manager  
Medium & Light Rotorcraft





**Associated Technical Documentation:**

Certification Plan COMPLIANCE PROGRAM 'CHANGES TO SEAT CUSHIONS/COVERS, AND NEW UPHOLSTERY COVERS ON ROBINSON R44/R44II/R66' of 26 July 2022 signed 26 August 2022.

Note: The certification plan covers two different helicopter types based on the commonality between both types and equal or stricter requirements met for both certification basis.

Or later revisions of the above listed document(s) approved/accepted under the EASA system.

**Limitations/Conditions:**

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -